

SIDDHARTH INSTITUTE OF ENGINEERING & TECHNOLOGY:: PUTTUR
(AUTONOMOUS)

M.Tech I Year I Semester Regular & Supplementary Examinations February-2025

ADVANCED SOLID MECHANICS

(Structural Engineering)

Time: 3 Hours

Max. Marks: 60

(Answer all Five Units 5 x 12 = 60 Marks)

UNIT-I

- | | | | | | |
|---|---|--|------------|-----------|-----------|
| 1 | a | Derive the differential equation of equilibrium in terms of displacement components for plane stress problem in the presence of body forces. | CO4 | L2 | 6M |
| | b | Explain plane stress and plane strain with examples. | CO4 | L2 | 6M |

OR

- | | | | | |
|---|--|------------|-----------|------------|
| 2 | List the six components of strain. Derive the strain components between the same for the different planes. | CO4 | L3 | 12M |
|---|--|------------|-----------|------------|

UNIT-II

- | | | | | |
|---|--|------------|-----------|------------|
| 3 | Determine the stress components and sketch their variation in a region included $y=0, y=d$ and $x=0$ on the side is positive. For the given stress function: | CO1 | L2 | 12M |
|---|--|------------|-----------|------------|

$$\Phi = \frac{-F}{d^3} xy^2 (3d - 2y)$$

OR

- | | | | | |
|---|---|------------|-----------|------------|
| 4 | A cantilever of length 'L' and depth 2C is of unit thickness. A force of P is applied at the free end. The upper and the lower edges are free from load. Obtain the equation of deflection curve of the beam in the form Where X is the distance from free end. | CO3 | L3 | 12M |
|---|---|------------|-----------|------------|

$$(V)_{Y=0} = \frac{PX^3}{6EI} - \frac{PL^2X}{2EI} + \frac{PL^3}{3EI}$$

UNIT-III

- | | | | | |
|---|--|-----------|------------|------------|
| 5 | Derive the stress components of a plate with circular hole subjected to uniaxial load. | L2 | CO1 | 12M |
|---|--|-----------|------------|------------|

OR

- | | | | | |
|---|---|-----------|------------|------------|
| 6 | Explain generalized solution of the two-dimensional problem in polar coordinates. | L2 | CO1 | 12M |
|---|---|-----------|------------|------------|

UNIT-IV

- 7 Determine the principal stress tensor at a point in a material if the strain tensor at a point is given below And Poisson's ratio 0.3. Define stress invariants also. **L2 CO2 12M**

$$\begin{bmatrix} +600 & -200 & +300 \\ -200 & +200 & +450 \\ +300 & -450 & -400 \end{bmatrix} \times 10^{-6} \quad E = 2 \times 10^5 \text{ N/mm}^2$$

OR

- 8 What are the stress invariants? Derive expression for the stress invariants. The state of stress is given at a point by following matrix. Determine principle stresses and principle directions. **CO2 L2 12M**

$$\begin{bmatrix} 9 & 6 & 3 \\ 6 & 5 & 2 \\ 3 & 2 & 4 \end{bmatrix}$$

UNIT-V

- 9 Explain and derive the equation for the Prandtl's membrane analogy **CO5 L2 12M**

OR

- 10 Explain the membrane analogy, applied to a narrow rectangular section. **CO5 L2 12M**

***** END *****